

Bird Strike Simulations On Composite Aircraft Structures

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{PDF} Bird Strike Simulations on Composite Aircraft

Bird impact on composite wing leading edge As a second example the bird strike simulation on a composite wing leading edge slat with Abaqus/Explicit is presented. The leading edge structure consists of a composite skin, five composite ribs and a metallic back plate, connected by rivets and adhesive bonding.

Bird Strike Simulations on Composite Aircraft Structures

The Johnson-Cook model was used for aluminum alloys LY-12 and 2024-T3 in [101] Simulations of bird strikes on composite plates can be found in [76, 87, 89,106,107]. With cohesive elements, Heimbs...

{PDF} Bird Strike Simulations on Composite Aircraft Structures

A methodology for the numerical simulation of bird strike on a novel leading edge (LE) structure of a horizontal tail plane is presented. The innovative LE design is based on the 'tensor skin' concept, comprising one or more folded composite sub-laminates that unfold during the bird impact, thus providing high-energy absorption characteristics.

Bird strike simulation on a novel composite leading edge

Research on Bird Strike Simulation of Composite Leading Edge Article (PDF Available) in AASRI Procedia 3:674-679 · December 2012 with 279 Reads How we measure 'reads'

{PDF} Research on Bird Strike Simulation of Composite

Bird Strike Simulation of Composite Aircraft Structure AMTAS New Project Proposal There is an increasing trend of birds colliding with aircraft. Aircraft are most susceptible to bird impacts during takeoff and landing. Typical impacts occur on components such as wing leading edges, radomes, turbofan engines, and cockpit windshields.

Bird Strike Simulation for Composite Aircraft Structure

In present simulations, E bird-strike approximately equals 29.65 kJ. In the considered impact duration, the total energy absorption (E absorption) of the impacted plate can be determined from the equation E absorption = $\sum N (1/2 m V_{relative}^2 - 1/2 m V_{residual}^2)$, where m is the mass of SPH particle; V residual is the residual velocity; and the symbol \sum depicts the sum of all SPH particles (N).

Bird Strike Resistance of Composite Laminates with

Abstract. A validated simulation methodology has been developed to support the bird-strike certification of the carbon fibre epoxy composite, moveable trailing edge (MTE) of the Boeing 787 Dreamliner. The explicit finite element software PAM-CRASH™ was selected to perform the simulations utilising the advanced composite material, fastener and smooth particle hydrodynamic bird models available in the code.

Bird strike simulation for certification of the Boeing 787

This phenomenon can be ascribed to the same structural configurations and the identical input impact energy (E impact) considered in present simulations. The bird-strike impact energy (E bird-strike) is defined by the impact velocity and projectile mass as E bird-strike = 0.5 × m bird × V relative 2, where m bird is the mass of bird and V relative is the relative velocity between the impacted structure and the projectile.

Bird Strike Resistance of Composite Laminates with

• Ellipsoidal bird recommended by International Bird Strike Group (ELSB) • Bird mass in all calculations was 0.68kg and length to diameter ratio was equal to two; Number of 9 Simulation Models Bird model - SPH U u 0.063 log 1.148 10 log 0.335 log 0.900m 10 10 Dm u particles Volume [10-3 m3] Diameter [m] Length [m]

Modelling of bird strike on the engine fan blades using FE-SPH

Highlights Bird strike simulations on aircraft structures have been performed and improved since the late 1970s. Large variety of bird impactor geometries, materials, masses, densities and modelling methods exists in the literature. Today, three established techniques for numerical bird impactor modelling are used: Lagrangian, Eulerian and SPH. Each technique has specific advantages ...

Computational methods for bird strike simulations: A

efficient simulation of bird-strike incidents on composite panels. 2. Manufacturing of flat un-stiffened and stiffened composite panels that include supporting stringers, sparcaps and other structural features. 3. Design and analysis of a supporting frame for bird-strike impact tests on composite panels. 4. Bird-strike tests and NDI analysis on composite panels. 5.

BirdStrike -CORDIS

Figure 3: Bird strike simulation on preloaded composite plate (v = 150 m/s) The bird impact simulation of the 32 g gelatine projectile with velocities up to 200 m/s on the unloaded plate led to no penetration but severe internal damage.

TOWARDS THE INDUSTRIAL ASSESSMENT OF BIRD STRIKE

Loads calculation, stress analysis and bird strike simulation of a composite wing leading edge Institute of Aerospace Engineering, Brno University of Technology 9 1. Introduction Composite structures are increasingly being used within the aircraft industry, even for primary structures.

LOADS CALCULATION, STRESS ANALYSIS AND BIRD STRIKE

Bird strike simulations are challenging because they are of short duration, cause large material deformation, and involve interactions between bodies with rapidly changing surfaces. The difficulty is increased by the need to model composite materials that include numerous layers, each with its own material, footprint, thickness and orientation.

To the Test - Volume X, Issue 2 | ANSYS

the composite structures modeling for bird strike phenomenon in order to validate available numerical models through full scale tests and simulation tools and also addresses a critical review on analysis techniques.

BIRD STRIKE SIMULATION ON COMPOSITE STRUCTURES

In this paper, the numerical methodologies, commonly adopted for the simulation of the bird strike event, are presented and assessed focusing on their capability to predict the induced damage and the composite components' residual strength.

Numerical methodologies for simulating bird strike on

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Simulating Bird Strike On Aircraft Composite Wing Leading

A bird strike event on an aircraft composite structure can be successfully simulated with ABAQUS/Explicit using CEL approach. With its strong composite damage and failure modeling capabilities and general contact algorithm, ABAQUS/Explicit is an ideal tool for such highly dynamic, nonlinear applications.

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